

Civil Aviation Organization Private Pilot Written Exam

Jan 2017



FIGURES

FIGURE P-01



FIGURE P-02



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FIGURE P-03

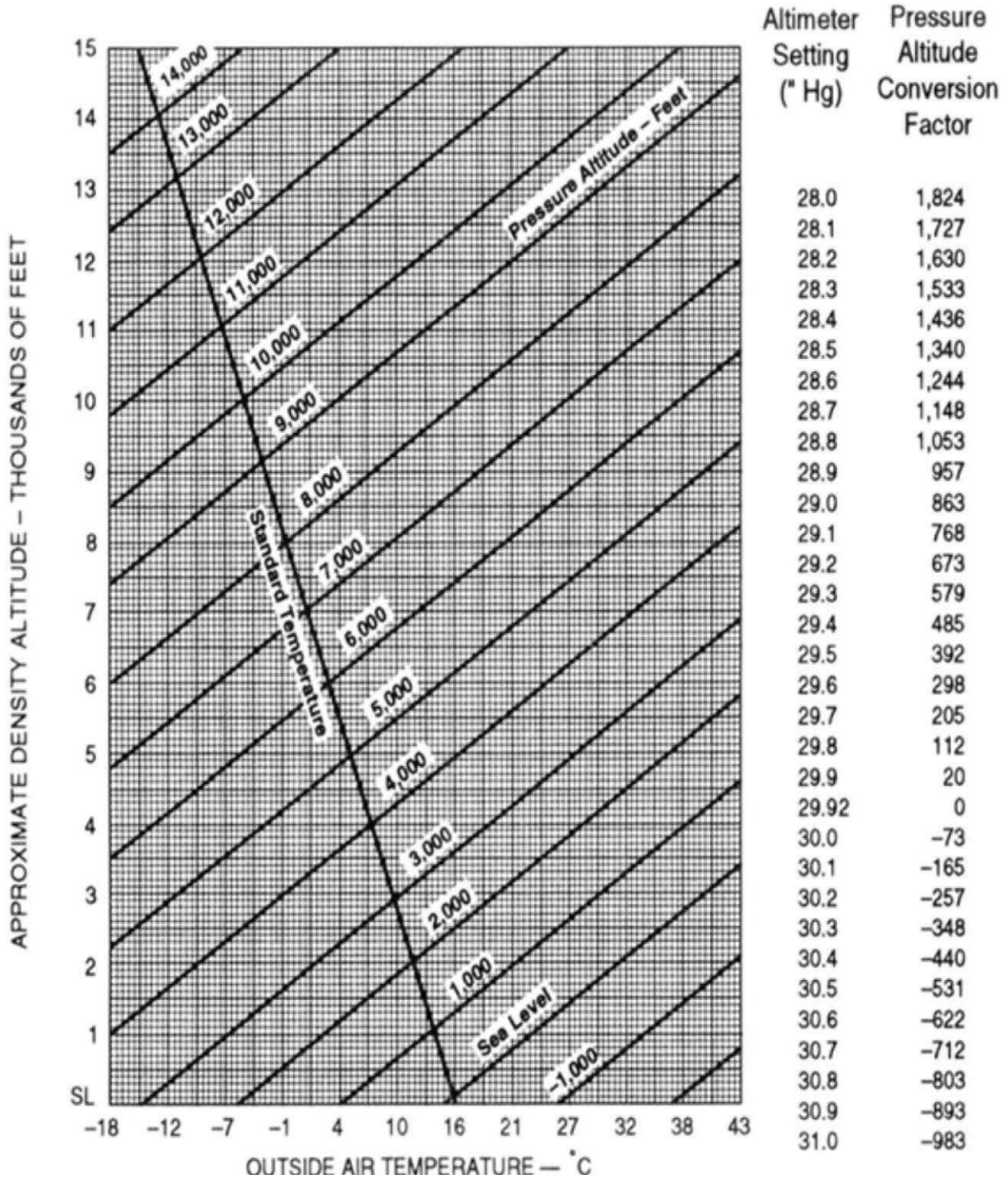


FIGURE P-04



DENSITY ALTITUDE CHART

FIGURE P-06



NORMAL CLIMB – 110 KIAS

FIGURE P-07

CONDITIONS:
Flaps Up
Gear Up
2500 RPM
30 Inches Hg
120 PPH Fuel Flow
Cowl Flaps Open
Standard Temperature

NOTES:

1. Add 16 pounds of fuel for engine start, taxi and takeoff allowance.
2. Increase time, fuel and distance by 10% for each 7 °C above standard temperature.
3. Distances shown are based on zero wind.

WEIGHT LBS	PRESS ALT FT	RATE OF CLIMB FPM	FROM SEA LEVEL		
			TIME MIN	FUEL USED POUNDS	DISTANCE NM
4000	S.L.	605	0	0	0
	4000	570	7	14	13
	8000	530	14	28	27
	12,000	485	22	44	43
	16,000	430	31	62	63
	20,000	365	41	82	87
3700	S.L.	700	0	0	0
	4000	665	6	12	11
	8000	625	12	24	23
	12,000	580	19	37	37
	16,000	525	26	52	53
	20,000	460	34	68	72
3400	S.L.	810	0	0	0
	4000	775	5	10	9
	8000	735	10	21	20
	12,000	690	16	32	31
	16,000	635	22	44	45
	20,000	565	29	57	61

FIGURE P-08

CRUISE POWER SETTINGS
65% MAXIMUM CONTINUOUS POWER (OR FULL THROTTLE)
2800 POUNDS

PRESS ALT	ISA -20 °C (-36 °F)							STANDARD DAY (ISA)							ISA +20 °C (+36 °F)									
	IOAT		ENGINE SPEED	MAN. PRESS	FUEL FLOW PER ENGINE		TAS	IOAT		ENGINE SPEED	MAN. PRESS	FUEL FLOW PER ENGINE		TAS	IOAT		ENGINE SPEED	MAN. PRESS	FUEL FLOW PER ENGINE		TAS			
	FEET	°F	°C	RPM	IN HG	PSI	GPH	KTS	MPH	°F	°C	RPM	IN HG	PSI	GPH	KTS	MPH	°F	°C	RPM	IN HG	PSI	GPH	KTS
SL	27	-3	2450	20.7	6.6	11.5	147	169	63	17	2450	21.2	6.6	11.5	150	173	99	37	2450	21.8	6.6	11.5	153	176
2000	19	-7	2450	20.4	6.6	11.5	149	171	55	13	2450	21.0	6.6	11.5	153	176	91	33	2450	21.5	6.6	11.5	156	180
4000	12	-11	2450	20.1	6.6	11.5	152	175	48	9	2450	20.7	6.6	11.5	156	180	84	29	2450	21.3	6.6	11.5	159	183
6000	5	-15	2450	19.8	6.6	11.5	155	178	41	5	2450	20.4	6.6	11.5	158	182	79	26	2450	21.0	6.6	11.5	161	185
8000	-2	-19	2450	19.5	6.6	11.5	157	181	36	2	2450	20.2	6.6	11.5	161	185	72	22	2450	20.8	6.6	11.5	164	189
10000	-8	-22	2450	19.2	6.6	11.5	160	184	28	-2	2450	19.9	6.6	11.5	163	188	64	18	2450	20.3	6.5	11.4	166	191
12000	-15	-26	2450	18.8	6.4	11.3	162	186	21	-6	2450	18.8	6.1	10.9	163	188	57	14	2450	18.8	5.9	10.6	163	188
14000	-22	-30	2450	17.4	5.8	10.5	159	183	14	-10	2450	17.4	5.6	10.1	160	184	50	10	2450	17.4	5.4	9.8	160	184
16000	-29	-34	2450	16.1	5.3	9.7	156	180	7	-14	2450	16.1	5.1	9.4	156	180	43	6	2450	16.1	4.9	9.1	155	178

NOTES: 1. Full throttle manifold pressure settings are approximate.
2. Shaded area represents operation with full throttle.

FIGURE P-09

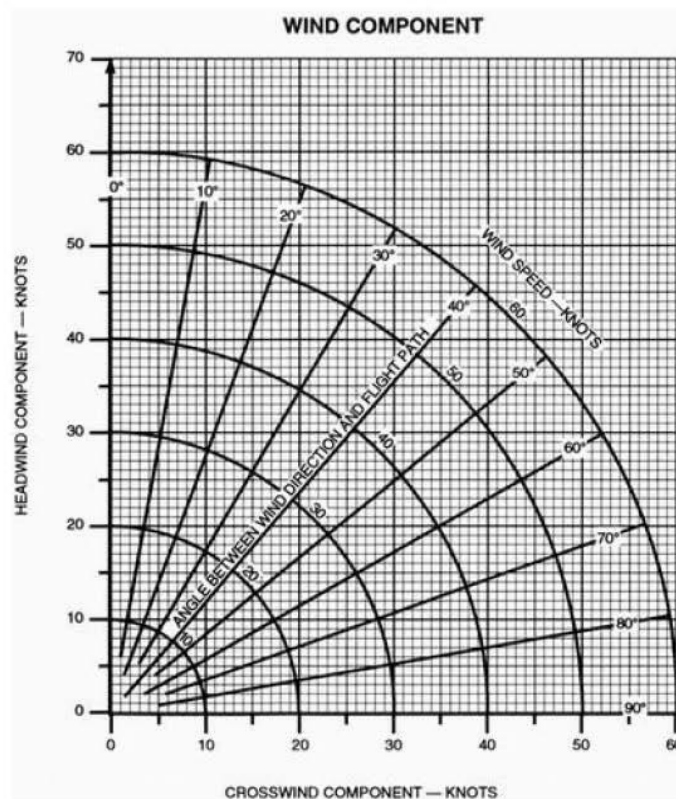


FIGURE P-10

LANDING DISTANCE

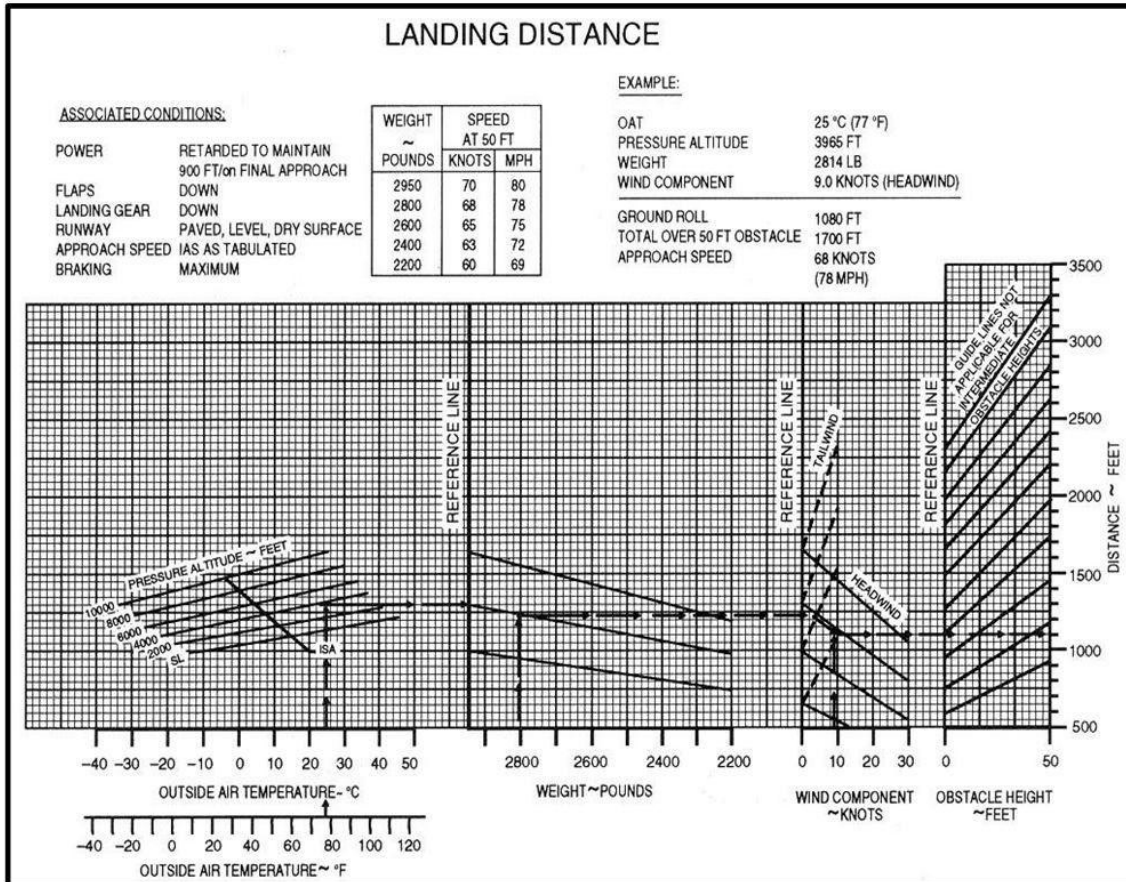


FIGURE P-10-A

LANDING DISTANCE		FLAPS LOWERED TO 40 ° - POWER OFF HARD SURFACE RUNWAY - ZERO WIND							
GROSS WEIGHT LB	APPROACH SPEED, IAS, MPH	AT SEA LEVEL & 59 °F		AT 2500 FT & 50 °F		AT 5000 FT & 41 °F		AT 7500 FT & 32 °F	
		GROUND ROLL	TOTAL TO CLEAR 50 FT OBS	GROUND ROLL	TOTAL TO CLEAR 50 FT OBS	GROUND ROLL	TOTAL TO CLEAR 50 FT OBS	GROUND ROLL	TOTAL TO CLEAR 50 FT OBS
1600	60	445	1075	470	1135	495	1195	520	1255

NOTES: 1. Decrease the distances shown by 10% for each 4 knots of headwind.
2. Increase the distance by 10% for each 60 °F temperature increase above standard.
3. For operation on a dry, grass runway, increase distances (both "ground roll" and "total to clear 50 ft obstacle") by 20% of the "total to clear 50 ft obstacle" figure.

Figure P-11

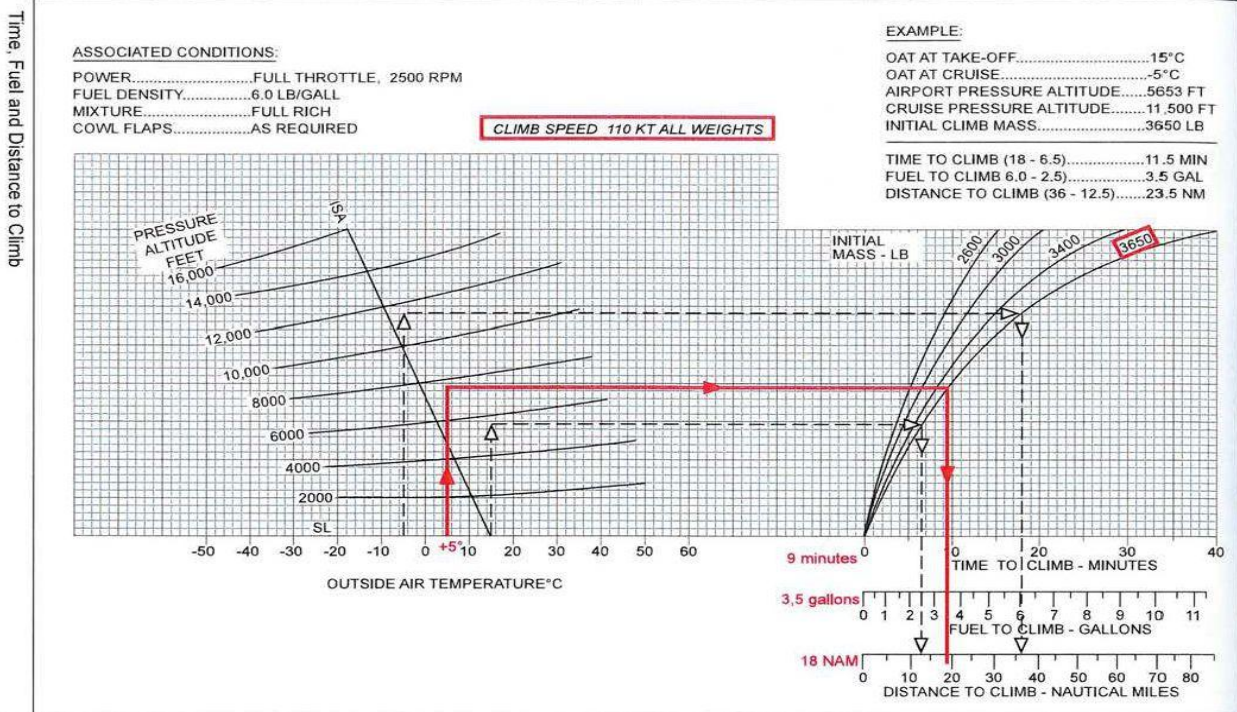


Figure P-12

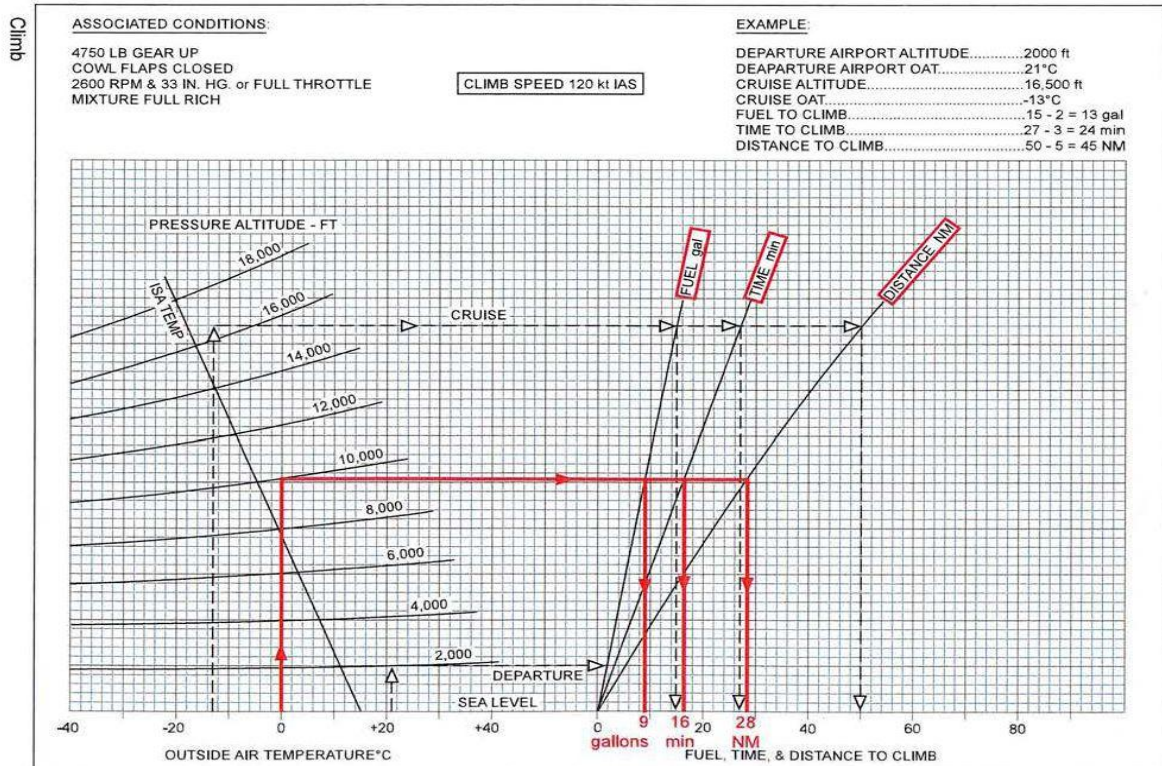


Figure P-13

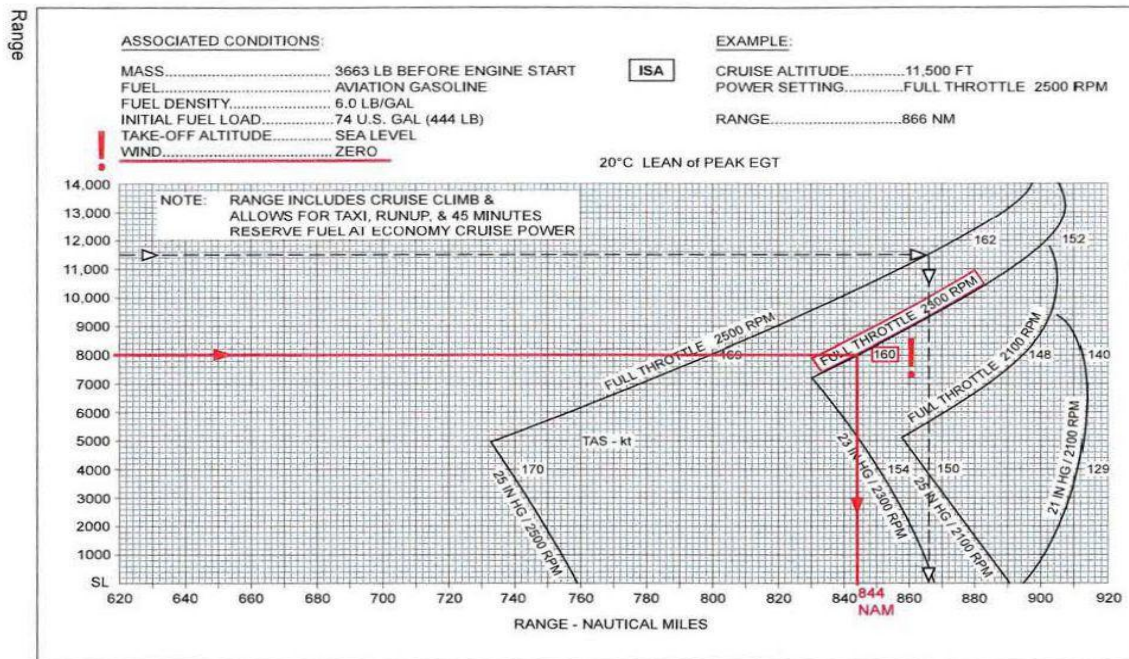


Figure P-14

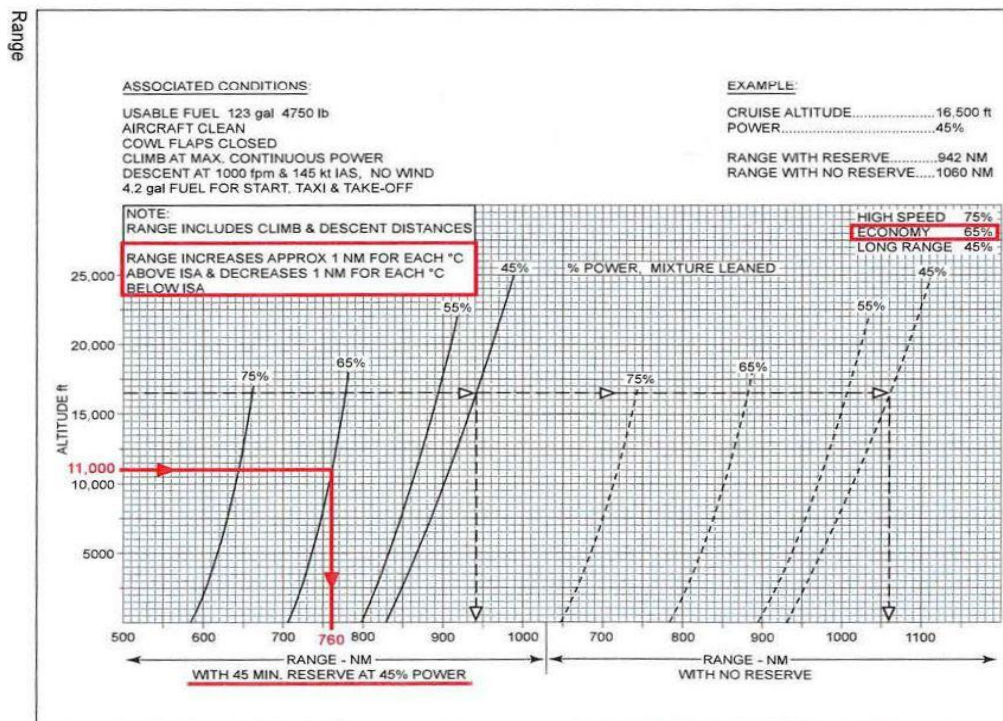


Figure P-15

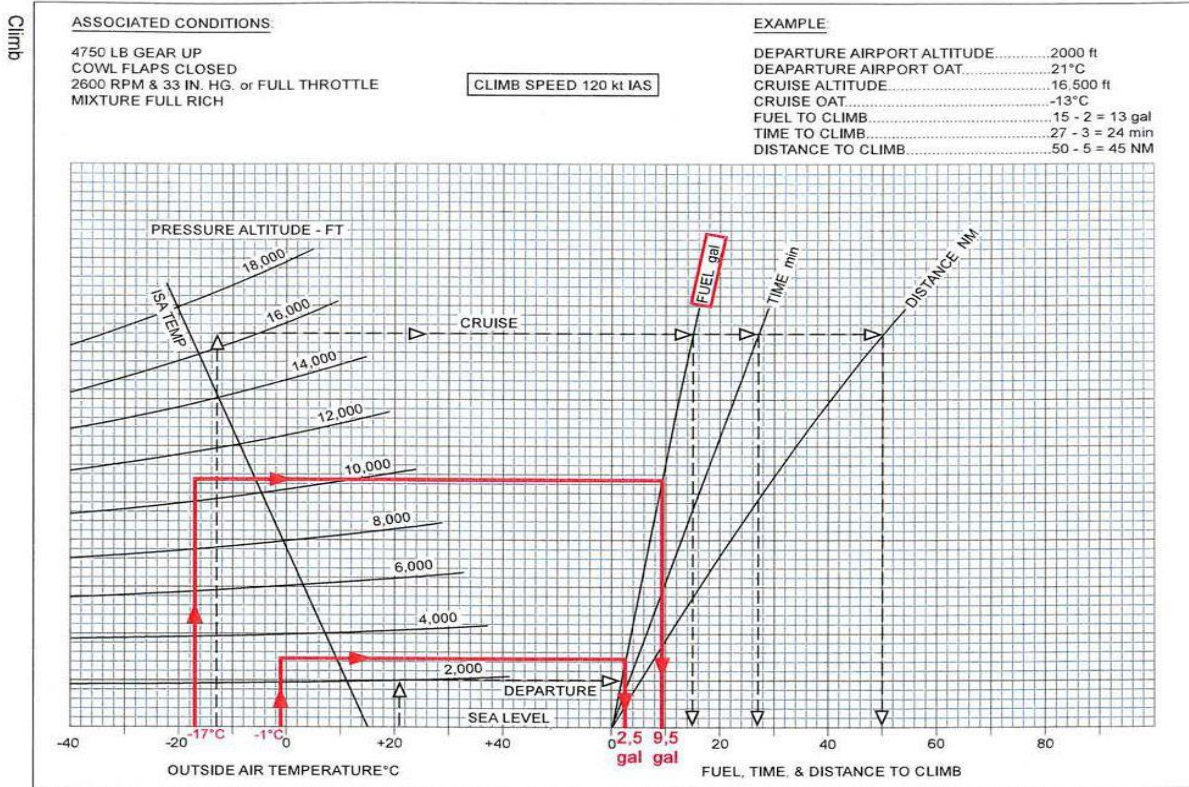


Figure P-16

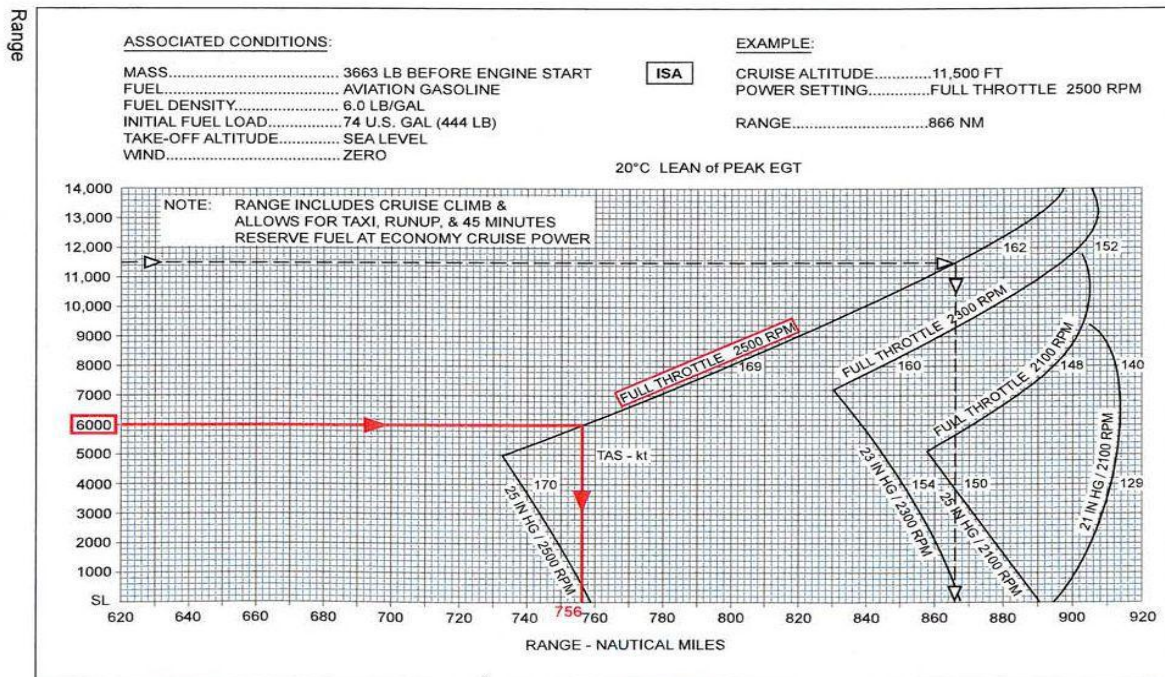


Figure P-17

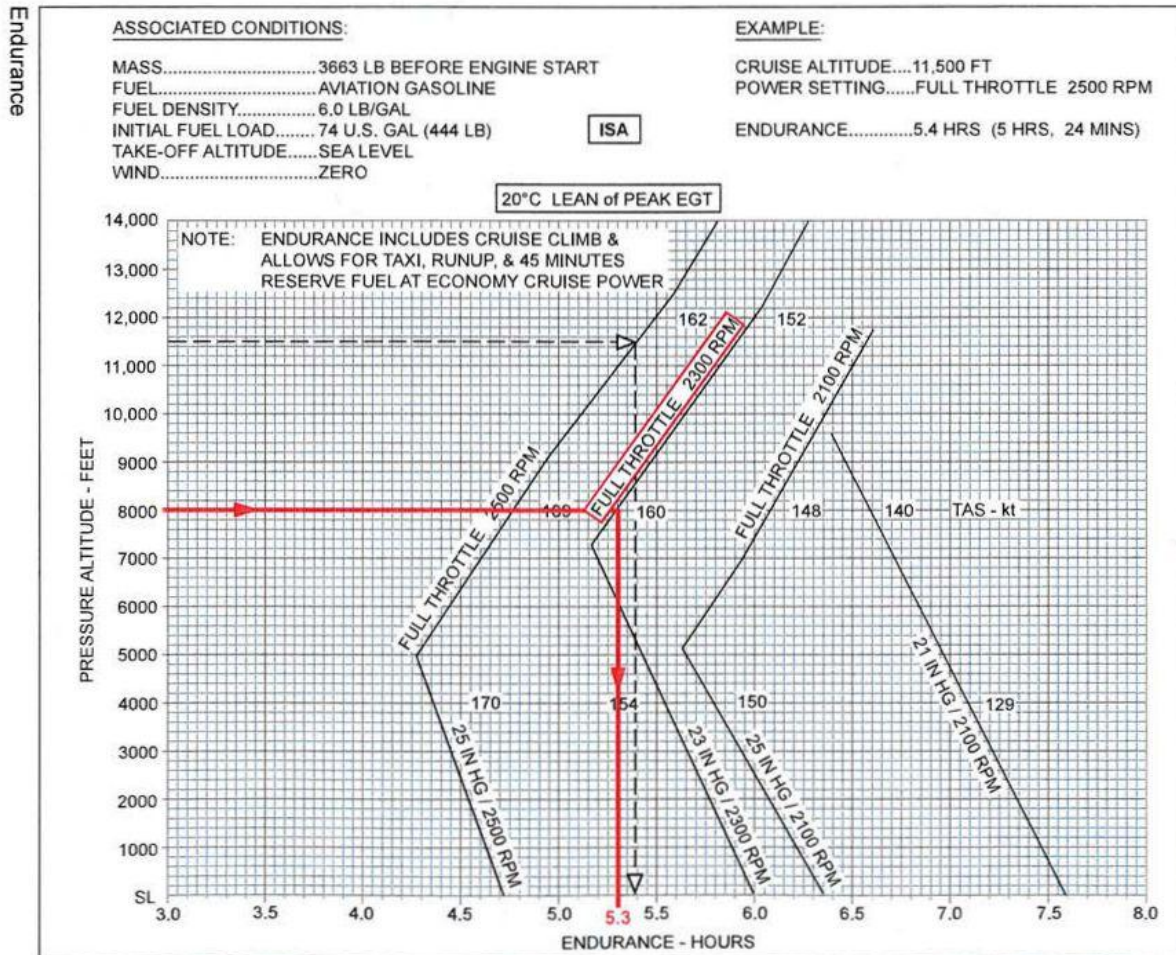


Figure P-18

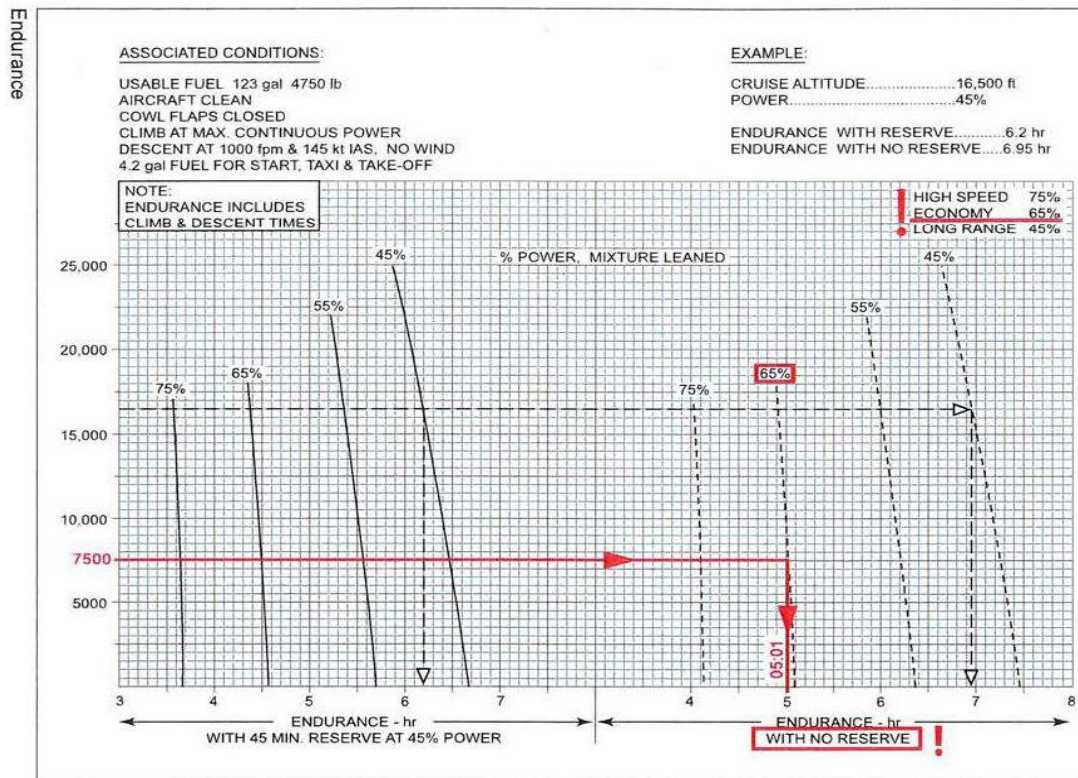


Figure P-19

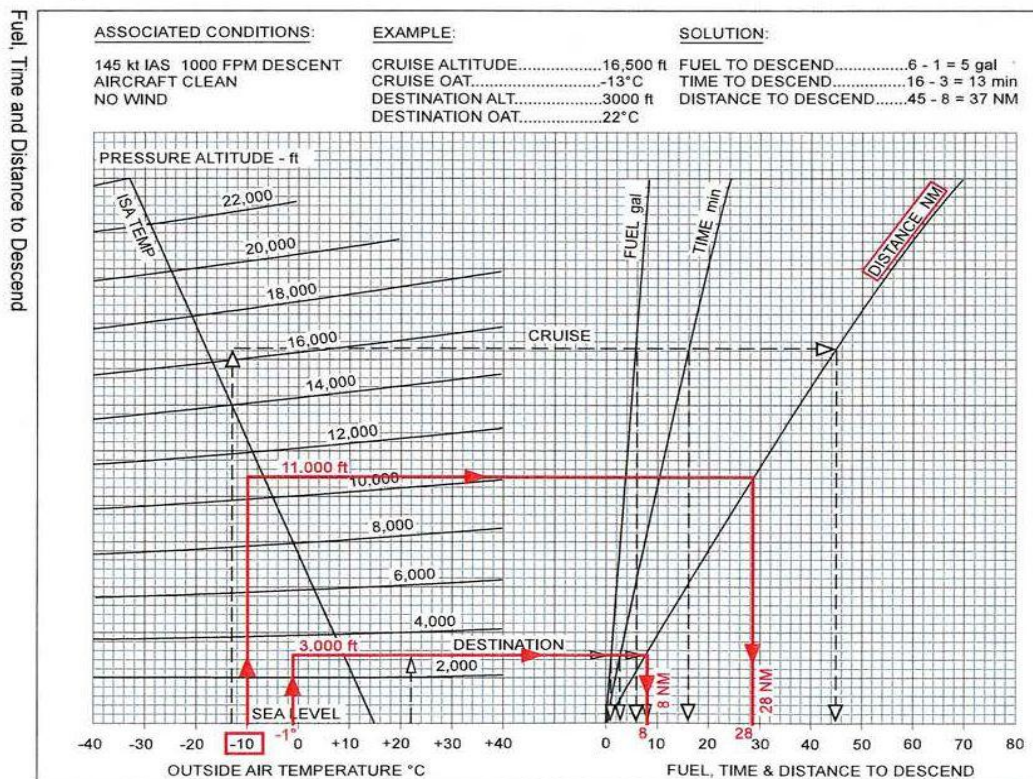


Figure P-20

Speed v Power

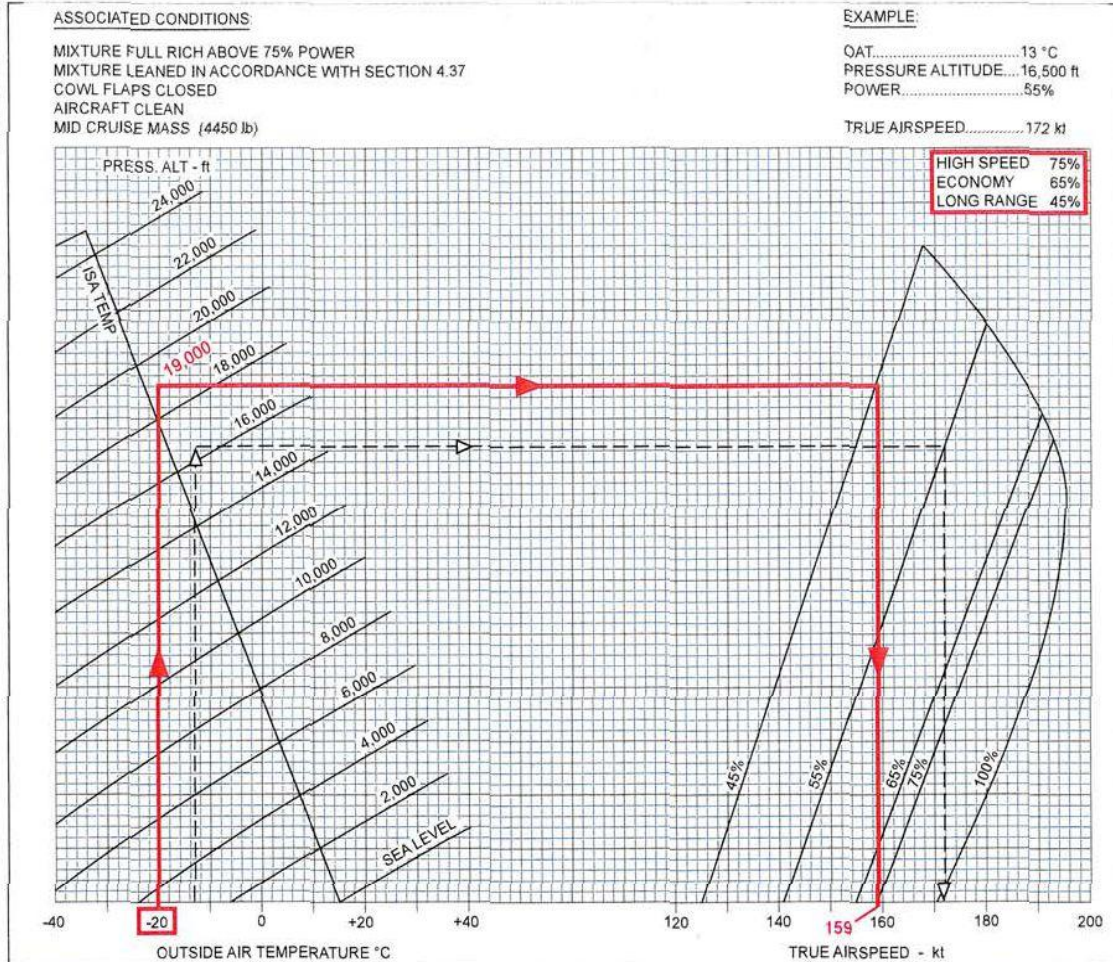


Figure P-21

Table 2.2.3 23.0 in. Hg (or full throttle) @ 2,300 rpm
Off-peak EGT Cruise lean mixture @ cruise weight 3,400 lb

ISA Dev.	Press. Alt.	IOAT		Man. Press.	Fuel Flow		Airspeed	
		°C	°F		In. Hg	PPH	GPH	KIAS
-20	0	-3	26	23.0	67.6	11.3	152	144
	2,000	-7	20	23.0	69.7	11.6	152	149
	4,000	-11	13	23.0	72.1	12.0	153	154
	6,000	-15	6	23.0	74.4	12.4	153	158
	8,000	-18	-1	22.4	73.8	12.3	150	160
	10,000	-23	-9	20.7	68.4	11.4	143	157
	12,000	-27	-16	19.2	63.8	10.6	135	153
	14,000	-31	-23	17.8	60.0	10.0	127	148
0	16,000	-35	-31	16.4	56.3	9.4	117	141
	0	17	62	23.0	65.4	10.9	147	145
	2,000	13	56	23.0	67.4	11.2	147	149
	4,000	9	49	23.0	69.4	11.6	148	154
	6,000	5	42	23.0	71.7	12.0	148	159
	8,000	2	35	22.4	71.1	11.9	145	160
	10,000	-3	27	20.7	66.2	11.0	137	157
	12,000	-7	20	19.2	61.8	10.3	129	152
+20	14,000	-11	13	17.8	58.5	9.8	120	146
	16,000	-15	5	16.4	55.3	9.2	109	137
	0	37	98	23.0	63.2	10.5	142	145
	2,000	33	92	23.0	65.1	10.9	143	149
	4,000	29	85	23.0	67.1	11.2	143	154
	6,000	25	78	23.0	69.0	11.5	142	158
	8,000	22	71	22.4	68.5	11.4	140	160
	10,000	17	63	20.7	64.0	10.7	132	156
	12,000	13	56	19.2	60.0	10.0	123	151
	14,000	9	48	17.8	57.1	9.5	113	142
	16,000	-	-	-	-	-	-	-
	16,000	-	-	-	-	-	-	-

ISA + 0°C
7500 ft
11,975 GPH
159,75 KTAS

ISA + 20°C
7500 ft
11,475 GPH
159,5 KTAS

ISA + 10°C
7500 ft
11,7 GPH
159,62 KTAS

Recommended Cruise Power Settings (continued)

- NOTE 1: Full-throttle manifold pressure settings are approximate.
- NOTE 2: Shaded areas represent operation with full throttle.
- NOTE 3: Fuel flows are to be used for flight planning. Lean using the EGT.

Figure P-22

Table 2.2.3 **23.0 in. Hg (or full throttle) @ 2,300 rpm**
Off-peak EGT Cruise lean mixture @ cruise weight 3,400 lb

ISA Dev.	Press. Alt.	IOAT		Man. Press.	Fuel Flow		Airspeed	
		°C	°F		PPH	GPH	KIAS	KTAS
-20	0	-3	26	23.0	67.6	11.3	152	144
	2,000	-7	20	23.0	69.7	11.6	152	149
	4,000	-11	13	23.0	72.1	12.0	153	154
	6,000	-15	6	23.0	74.4	12.4	153	158
	8,000	-18	-1	22.4	73.8	12.3	150	160
	10,000	-23	-9	20.7	68.4	11.4	143	157
	12,000	-27	-16	19.2	63.8	10.6	135	153
	14,000	-31	-23	17.8	60.0	10.0	127	148
	16,000	-35	-31	16.4	56.3	9.4	117	141
0	0	17	62	23.0	65.4	10.9	147	145
	2,000	13	56	23.0	67.4	11.2	147	149
	4,000	9	49	23.0	69.4	11.6	148	154
	6,000	5	42	23.0	71.7	12.0	148	159
	8,000	2	35	22.4	71.1	11.9	145	160
	10,000	-3	27	20.7	66.2	11.0	137	157
	12,000	-7	20	19.2	61.8	10.3	129	152
	14,000	-11	13	17.8	58.5	9.8	120	146
	16,000	-15	5	16.4	55.3	9.2	109	137
+20	0	37	98	23.0	63.2	10.5	142	145
	2,000	33	92	23.0	65.1	10.9	143	149
	4,000	29	85	23.0	67.1	11.2	143	154
	6,000	25	78	23.0	69.0	11.5	142	158
	8,000	22	71	22.4	68.5	11.4	140	160
	10,000	17	63	20.7	64.0	10.7	132	156
	12,000	13	56	19.2	60.0	10.0	123	151
	14,000	9	48	17.8	57.1	9.5	113	142
	16,000	-	-	-	-	-	-	-

Recommended Cruise Power Settings (continued)

- NOTE 1: Full-throttle manifold pressure settings are approximate.
 NOTE 2: Shaded areas represent operation with full throttle.
 NOTE 3: Fuel flows are to be used for flight planning. Lean using the EGT.

Figure P-23

Table 2.2.2 **25.0 in. Hg (or full throttle) @ 2,100 rpm**
Off-peak EGT Cruise lean mixture @ cruise weight 3,400 lb

ISA Dev.	Press. Alt.	IOAT		Man. Press.	Fuel Flow		Airspeed	
		°C	°F		PPH	GPH	KIAS	KTAS
-20	0	-3	26	25.0	63.8	10.6	148	140
	2,000	-7	19	25.0	66.4	11.1	149	145
	4,000	-11	12	25.0	68.9	11.5	149	150
	6,000	-15	5	24.3	68.3	11.4	147	152
	8,000	-19	-2	22.5	63.9	10.7	139	148
	10,000	-23	-9	20.8	60.1	10.0	132	144
	12,000	-27	-17	19.3	56.7	9.5	123	139
	14,000	-31	-24	17.9	54.5	9.1	113	132
	16,000	-35	-32	16.5	52.2	8.7	95	114
0	0	17	62	25.0	61.9	10.3	143	140
	2,000	13	55	25.0	64.2	10.7	143	145
	4,000	9	48	25.0	66.6	11.1	144	150
	6,000	5	41	24.3	66.1	11.0	141	152
	8,000	1	34	22.5	61.9	10.9	134	148
	10,000	-3	27	20.8	58.5	9.8	126	143
	12,000	-7	19	19.3	55.6	9.3	116	136
	14,000	-11	12	17.9	53.5	8.9	103	125
	16,000	-	-	-	-	-	-	-
+20	0	37	98	25.0	60.1	10.0	138	140
	2,000	33	91	25.0	62.1	10.4	138	145
	4,000	29	84	25.0	64.4	10.7	139	150
	6,000	25	77	24.3	63.9	10.7	136	151
	8,000	21	70	22.5	60.2	10.0	128	147
	10,000	17	63	20.8	56.8	9.5	119	141
	12,000	13	55	19.3	54.5	9.1	108	131
	14,000	-	-	-	-	-	-	-
	16,000	-	-	-	-	-	-	-

Recommended Cruise Power Settings (continued)

- NOTE 1: Full-throttle manifold pressure settings are approximate.
 NOTE 2: Shaded areas represent operation with full throttle.
 NOTE 3: Fuel flows are to be used for flight planning. Lean using the EGT.

Figure P-24

	TAS *)	Fuel flow *)	Ground speed (kts)	Time (hrs)	Fuel used (l)
FL50	192	208	162	2,72	565
FL100	201	192	151	2,91	558
FL180	216	163	146	3,02	492

*) Figure must be interpolated from table.

GS = TAS - wind

Time = 440 NM / GS

Fuel = Time * Fuel flow

From table it is evident, that FL180 will offer the lowest fuel economy => best range performance.

Figure P-25

Endurance / Fuel Calculation

	Fuel (kg)	Time (hh:mm)
Trip Fuel	5.800	02:32
Contingency Fuel	290	00:07
Alternate Fuel	1.800	00:42
Final Reserve Fuel	1.325	00:30
Minimum T/O Fuel	9.215	
Extra Fuel	585	00:15
Actual T/O Fuel	9.800	
Taxi Fuel	200	
Ramp Fuel	10.000	

Figure P-26

POWER		75%					65%					55%					45%				
FUEL FLOW		29.0 GPH					23.3 GPH					18.7 GPH					16.0 GPH				
RPM		2,500	2,600	2,400	2,500	2,600	2,100	2,200	2,300	2,400	2,500	2,600	2,100	2,200	2,300	2,400	2,500	2,600			
PRESS ALT (ft)	ISA 0°C	MANIFOLD ABSOLUTE PRESSURE (Hg in) (MAP)																			
		0	15	34.0	33.0	33.8	32.0	31.0	31.2	30.3	29.4	28.2	27.2	26.3	27.1	26.4	25.5	24.3	23.3	22.5	
2,000	11	33.8	32.7	33.2	31.7	30.7	30.5	29.7	28.8	27.8	26.8	26.0	26.4	25.8	24.6	23.7	22.8	22.1			
4,000	7	33.6	32.4	32.8	31.5	30.5	30.0	29.2	28.3	27.4	26.4	25.6	25.8	25.0	24.0	23.2	22.3	21.8			
6,000	3	33.4	32.2	32.5	31.2	30.3	29.7	28.8	28.0	27.0	26.2	25.3	25.3	24.5	23.5	22.8	21.9	21.5			
8,000	-1	33.1	32.0	32.3	31.0	30.1	29.4	28.4	27.7	26.8	25.7	25.0	24.8	24.0	23.0	22.4	21.6	21.2			
10,000	-5	33.0	31.9	32.0	30.9	30.0	-	28.3	27.5	26.5	25.5	24.7	24.4	23.7	22.8	22.0	21.4	21.0			
12,000	-9	32.5	31.8	31.8	30.7	29.8	-	28.3	27.2	26.3	25.3	24.6	24.0	23.3	22.5	21.7	21.2	20.9			
14,000	-13	-	31.7	-	30.5	29.7	-	-	27.1	26.1	25.2	24.4	-	23.0	22.3	21.4	21.1	20.8			
16,000	-17	-	31.6	-	30.4	29.5	-	-	-	25.9	25.0	24.3	-	-	22.0	21.3	21.0	20.6			
18,000	-21	-	-	-	-	29.4	-	-	-	-	25.0	24.2	-	-	-	21.2	20.9	20.5			
20,000	-25	-	-	-	-	29.3	-	-	-	-	-	24.2	-	-	-	21.2	20.8	20.4			
22,000	-28	-	-	-	-	-	-	-	-	-	-	24.1	-	-	-	-	-	20.4			
MAX EGT		1,525°F					1,650°F														
24,000	-33	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	20.4			
25,000	-34	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	20.4			

Power Setting Table

Figure P-27

Mass definitions	Airframe + Engines	Equipment (all roles)	Unusable fuel + Oil + Hydraulic fluid	Crew + Catering	Payload	Fuel
Basic empty mass (BEM)	Yes	Yes	Yes			
Dry operation mass (DOM)	Yes	Yes	Yes	Yes		
Zero fuel mass (ZMF)	Yes	Yes	Yes	Yes	Yes	
Ramp mass (RM)	Yes	Yes	Yes	Yes	Yes	Yes (Fuel Load)
Take off mass (TOM)	Yes	Yes	Yes	Yes	Yes	Yes (Toff fuel)
Operating mass (OM)	Yes	Yes	Yes	Yes	Yes	Yes (Toff fuel)
Gross mass (GM)	Yes	Yes	Yes	Yes	Yes	Yes (Fuel remaining)
Landing mass (LM)	Yes	Yes	Yes	Yes	Yes	Yes

Figure P-28

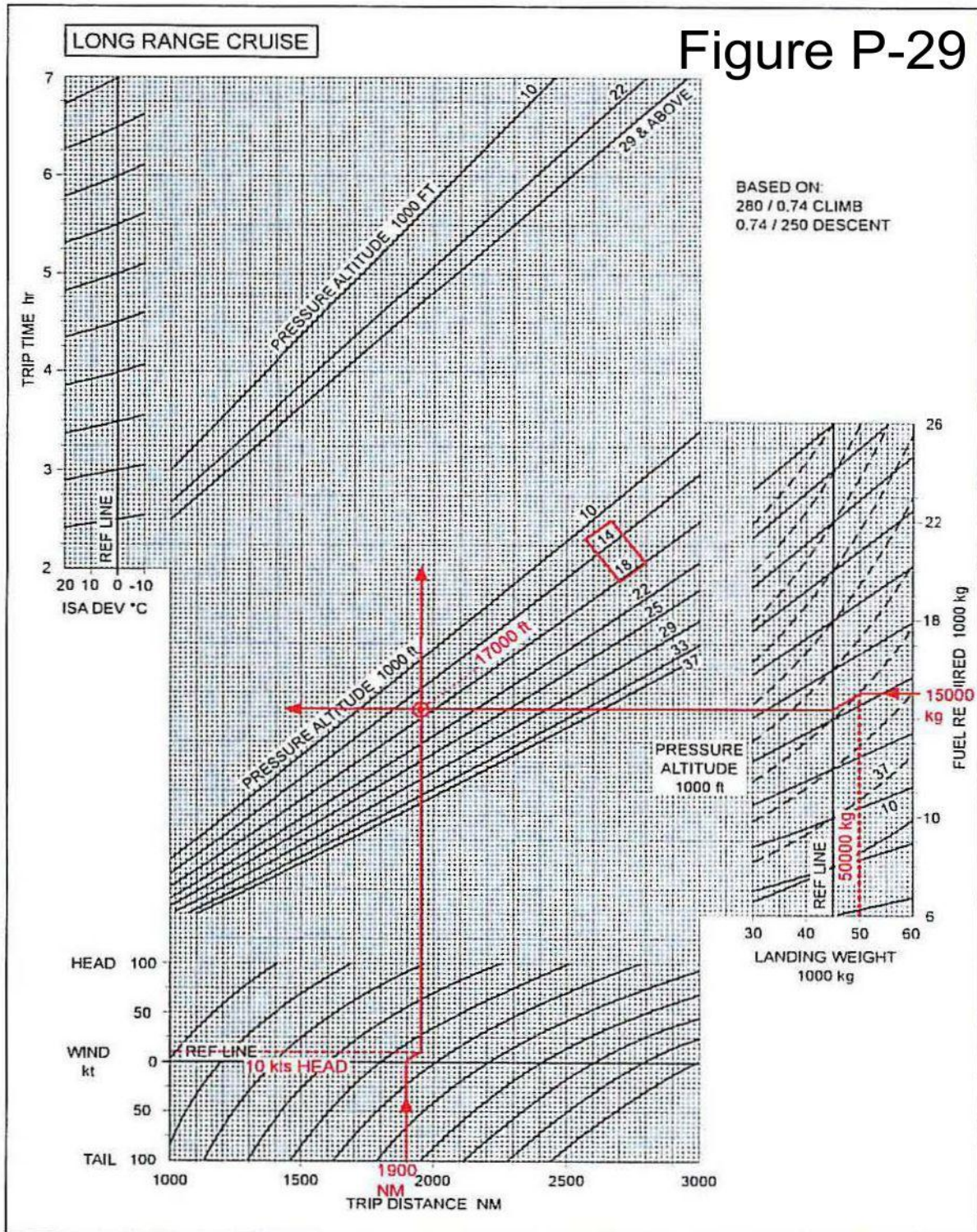
Table 2.2.3 **23.0 in. Hg (or full throttle) @ 2,300 rpm**
Off-peak EGT Cruise lean mixture @ cruise weight 3,400 lb

STANDARD DAY =
= 0° ISA DEVIATION

ISA Dev.	Press. Alt.	IOAT		Man. Press.	Fuel Flow		Airspeed	
		°C	°F		PPH	GPH	KIAS	KTAS
-20	0	-3	26	23.0	67.6	11.3	152	144
	2,000	-7	20	23.0	69.7	11.6	152	149
	4,000	-11	13	23.0	72.1	12.0	153	154
	6,000	-15	6	23.0	74.4	12.4	153	158
	8,000	-18	-1	22.4	73.8	12.3	150	160
	10,000	-23	-9	20.7	68.4	11.4	143	157
	12,000	-27	-16	19.2	63.8	10.6	135	153
	14,000	-31	-23	17.8	60.0	10.0	127	148
16,000	-35	-31	16.4	56.3	9.4	117	141	
0	0	17	62	23.0	65.4	10.9	147	145
	2,000	13	56	23.0	67.4	11.2	147	149
	4,000	9	49	23.0	69.4	11.6	148	154
	6,000	5	42	23.0	71.7	12.0	148	159
	8,000	2	35	22.4	71.1	11.9	145	160
	10,000	-3	27	20.7	66.2	11.0	137	157
	12,000	-7	20	19.2	61.8	10.3	129	152
	14,000	-11	13	17.8	58.5	9.8	120	146
16,000	-15	5	16.4	55.3	9.2	109	137	
+20	0	37	98	23.0	63.2	10.5	142	145
	2,000	33	92	23.0	65.1	10.9	143	149
	4,000	29	85	23.0	67.1	11.2	143	154
	6,000	25	78	23.0	69.0	11.5	142	158
	8,000	22	71	22.4	68.5	11.4	140	160
	10,000	17	63	20.7	64.0	10.7	132	156
	12,000	13	56	19.2	60.0	10.0	123	151
	14,000	9	48	17.8	57.1	9.5	113	142
16,000	-	-	-	-	-	-	-	

Recommended Cruise Power Settings (continued)

- NOTE 1: Full-throttle manifold pressure settings are approximate.
 NOTE 2: Shaded areas represent operation with full throttle.
 NOTE 3: Fuel flows are to be used for flight planning. Lean using the EGT.



Simplified Flight Planning – Trip Distances 1,000 NM to 3,000 NM

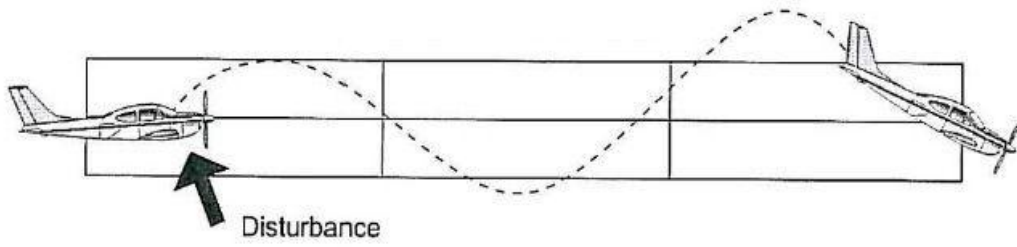


Figure P-30

Figure P-31

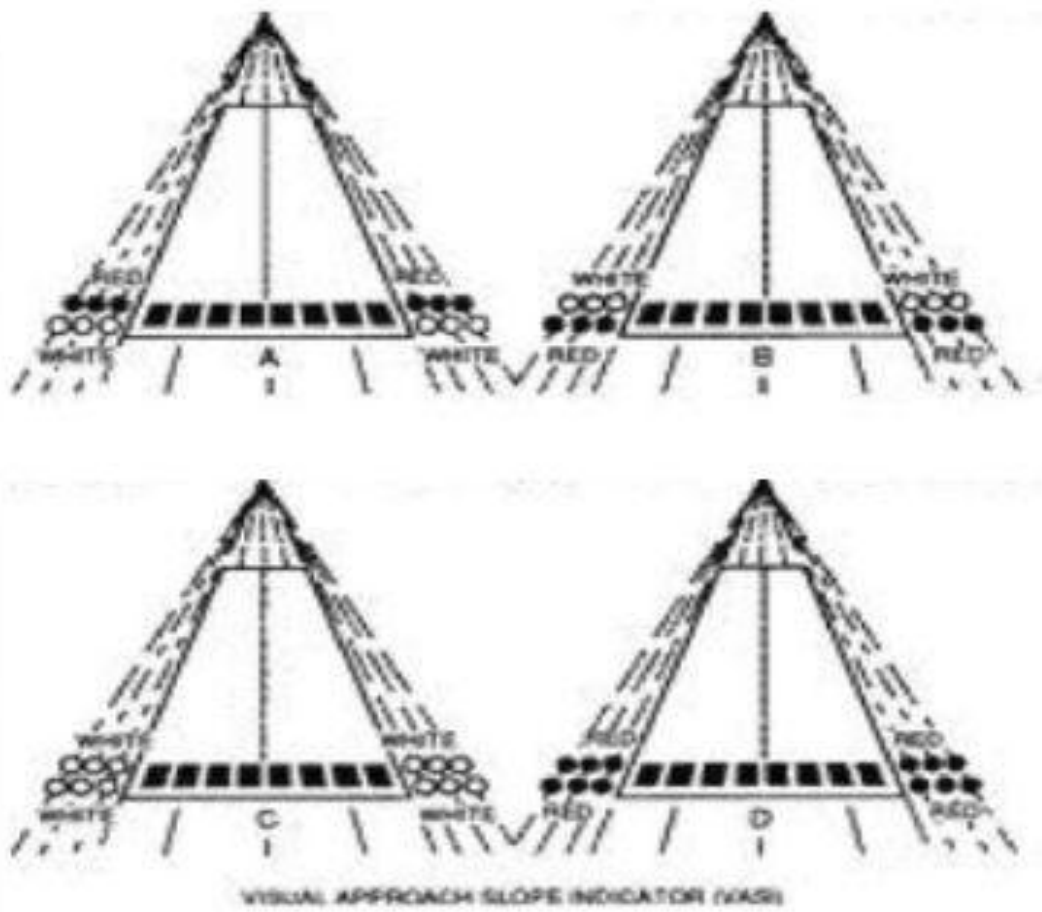


Figure P-32

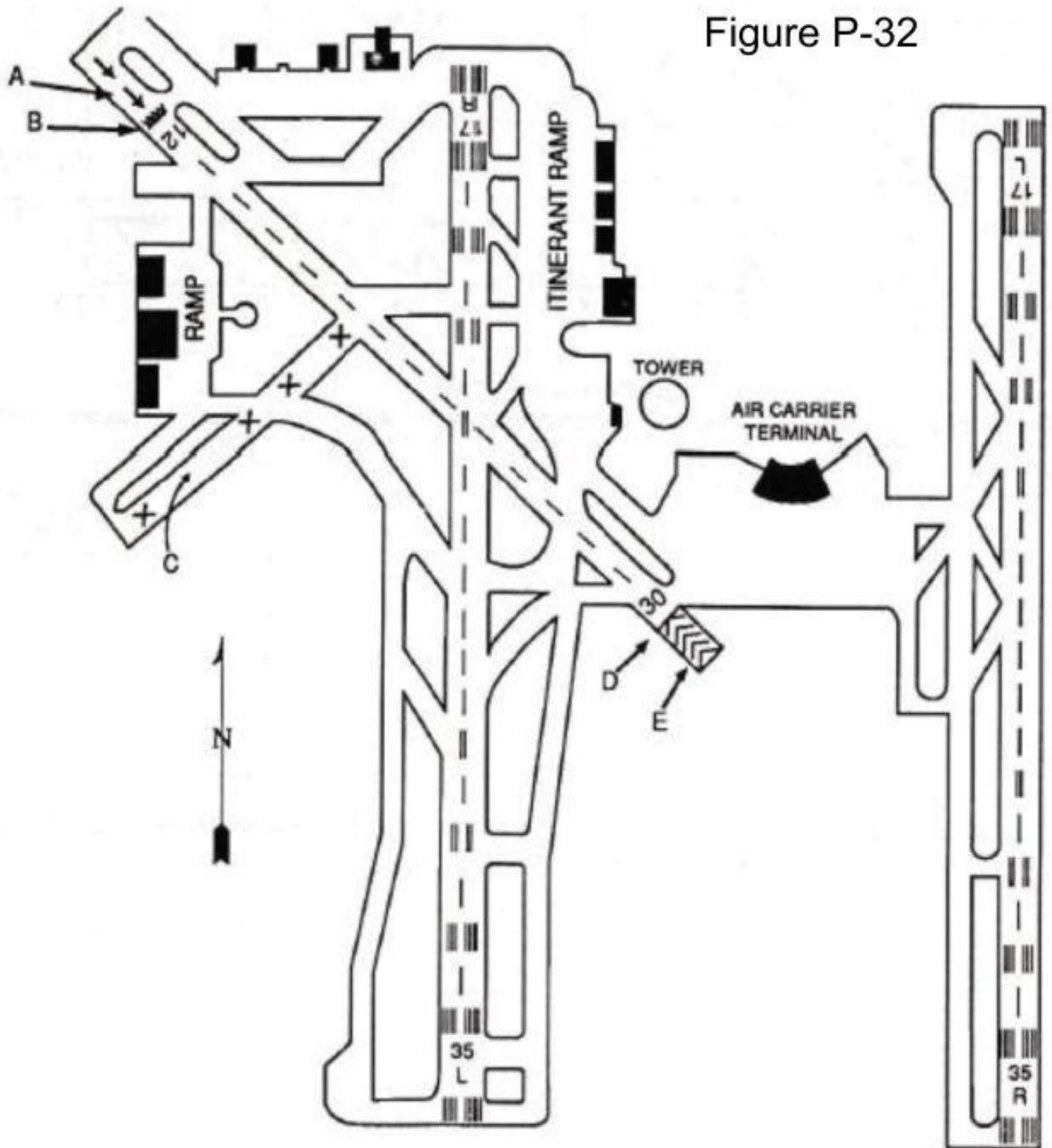


Figure P-33

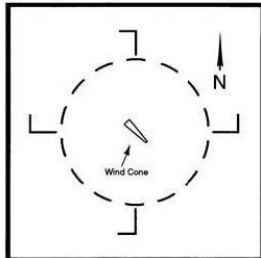
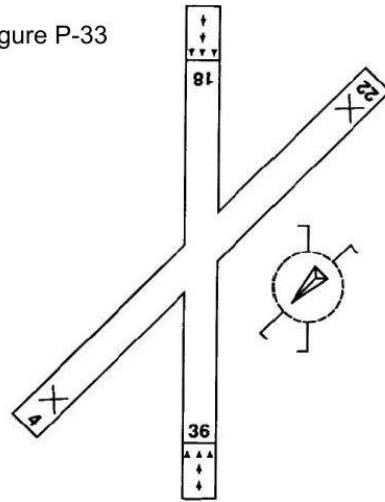


Figure P-34

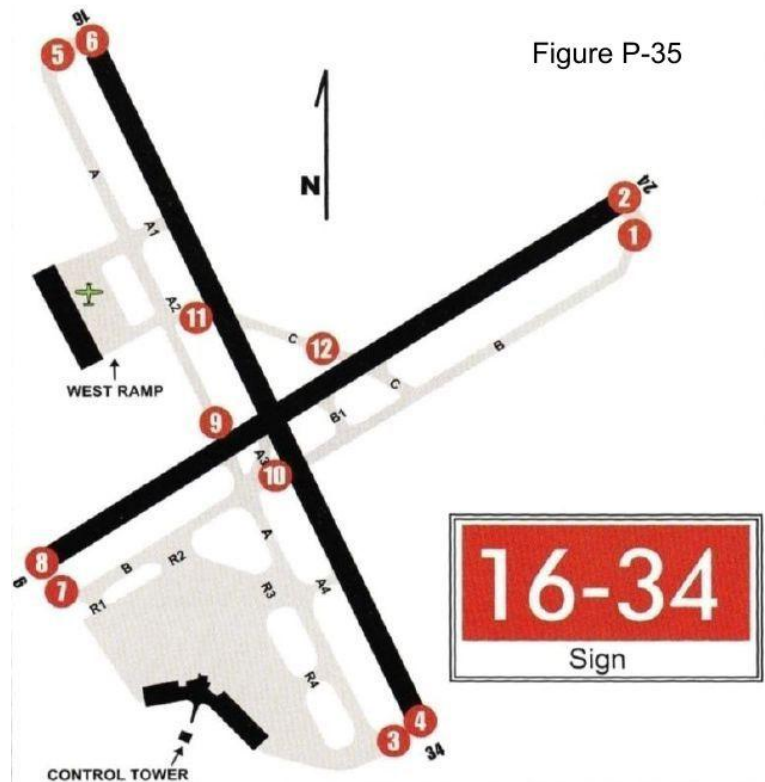


Figure P-35

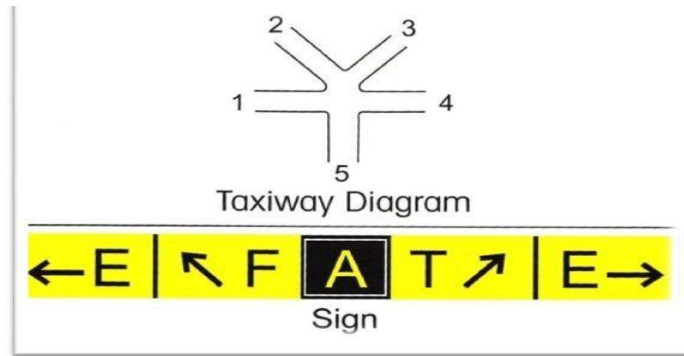


Figure P-36



Figure P-37



Figure P-38

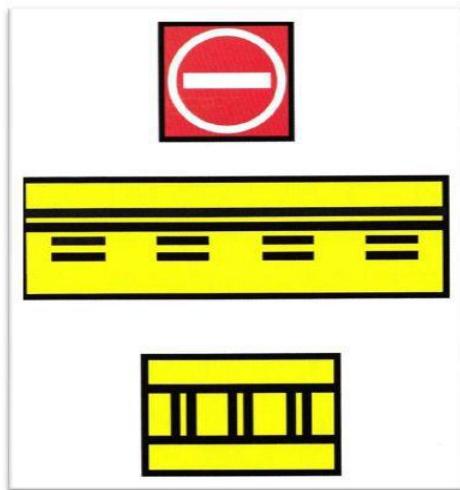


Figure P-39

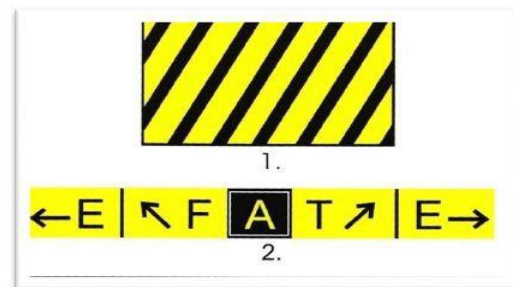


Figure P-40

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